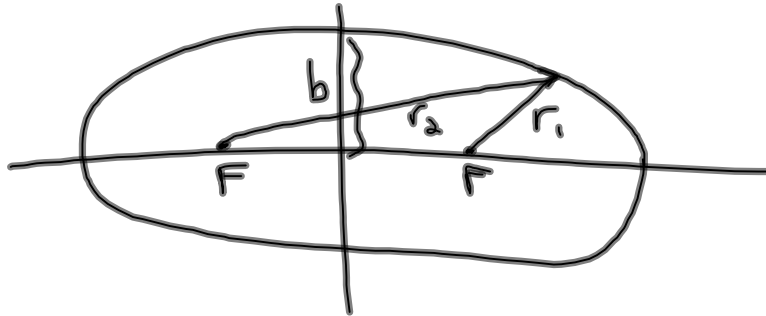


Gravity

Kepler's Laws (3)

- 1) All planets move in an elliptical orbit with the Sun at one of the foci



For planets -

perihelion \rightarrow point closest to the sun

aphelion \rightarrow point furthest from the sun

Semimajor = Average distance

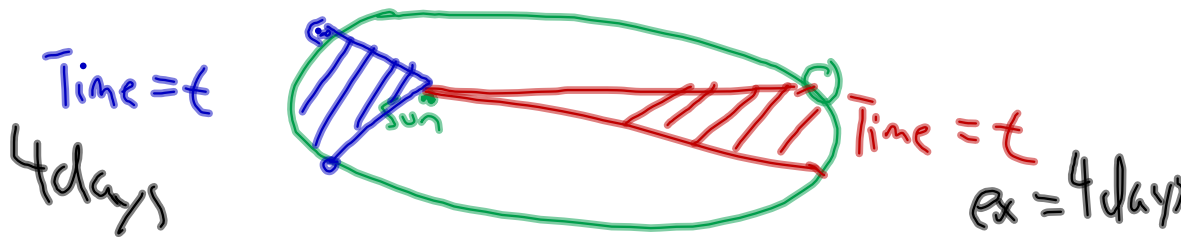
For earth from sun

$$1.5 \times 10^{11} \text{ m} = 1 \text{ A.U.} = (1 \text{ amu})$$

93.0 $\times 10^6$ Miles Astronomical Unit (Astronomical mean unit)

Kepler's 2nd Law -

The area swept out between a planet and the sun during its orbit is always equal for equal times.



3rd Law

The square of the period of any planet is proportional to the cube of the semimajor axis.

$$T^2 = Cr^3$$

C = constant for all planets

$$T^2 = Cr^3$$

For Neptune $164.8^2 \text{ yrs} = Cr^3$

For \oplus $1 \text{ yr}^2 = Cr^3$ $\frac{164.8^2}{1^2} = \frac{r^3}{1^3}$
 $1 \text{ yr}^2 = C 1 \text{ AU}^3$

$$164.8^2 = r^3$$

$$r = 164.8^{2/3}$$

$$r = 30.06 \text{ A.U.}$$

$$4.5 \times 10^{12} \text{ m}$$

$$r_e = 6371 \text{ km} = 6.37 \times 10^6 \text{ m}$$

$$M_e = 5.98 \times 10^{24} \text{ kg}$$

Universal Law of gravitation -

- Force due to gravity from object 1 on object 2 is

$$\vec{F} = - \frac{G m_1 m_2}{r_{12}^2} \hat{r}_{12}$$

r_{12} = distance between 1 and 2

\hat{r}_{12} = unit vector to indicated direction

G = universal gravitation constant


$$= 6.67 \times 10^{-11} \text{ Nm}^2/\text{kg}^2$$

$$F = G \frac{m_1 m_2}{r^2} \leftarrow \text{magnitude}$$

you as m_1
Earth as m_2

$$\frac{1}{r^2} = \frac{(6.67 \times 10^{-11}) \text{ Nm}^2}{(6.4 \times 10^6)^2} \times 5.98 \times 10^{24} \text{ kg}$$

For near circular orbits
 special case of Kepler's Laws.



$$F = \frac{G m_1 m_2}{r^2}$$

$m_1 = \text{Mass Sun}$
 $= M_s$

$m_2 = \text{Mass of Planet in "circular" orbit}$

$$F = \frac{G M_s M_p}{r^2} = M_p a_p = M_p \frac{v^2}{r}$$

$$v^2 = \frac{G M_s}{r} = \frac{4\pi^2 r^2}{T^2}$$

$$v = \frac{2\pi r}{T}$$

$$T^2 = \frac{4\pi^2 r^3}{G M_s}$$

$$T^2 = \left(\frac{4\pi^2}{G M_s} \right) r^3$$

for circular orbits, $C = \frac{4\pi^2}{G M_s}$

- Used for satellites orbiting planets. (natural (ie. moons) + others)

M_s would be mass of planet

$$dU = -F \cdot d\ell$$

$$dU = -\vec{F}_g \cdot d\ell = -(-F_g \hat{r}) d\ell$$

$$dU = \frac{GM_em}{r^2} dr = -\frac{U(r)}{r} dr$$

$$U = GM_em \int r^{-2} dr$$

assuming $U_0 = 0$ at $r = r_0$ for reference level

$$U = -\frac{GM_em}{r}$$

!

escape velocity

$$v_e = \sqrt{\frac{2GM_E}{R_E}}$$

M_E, R_E
for earth

$$6.67 \times 10^{-11} \text{ Nm}^2/\text{kg}^2$$

apply for whatever
planet
look

Gravitational Fields

$$F = ma \quad a = g(r)$$

$$F_g = mg(r)$$

$$g(r) = \frac{F_g}{m} = \frac{Gm_1m_2}{r^2m_1}$$

$$g(r) = \frac{G M_2 \leftarrow \text{Mass of planet}}{r^2}$$

→ Gravitational fields.

